

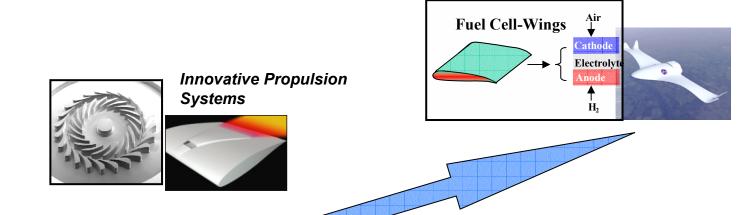
AT Vehicle Systems Program



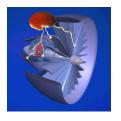


Propulsion & Power

Transport Electric Propulsion



Intelligent Engine







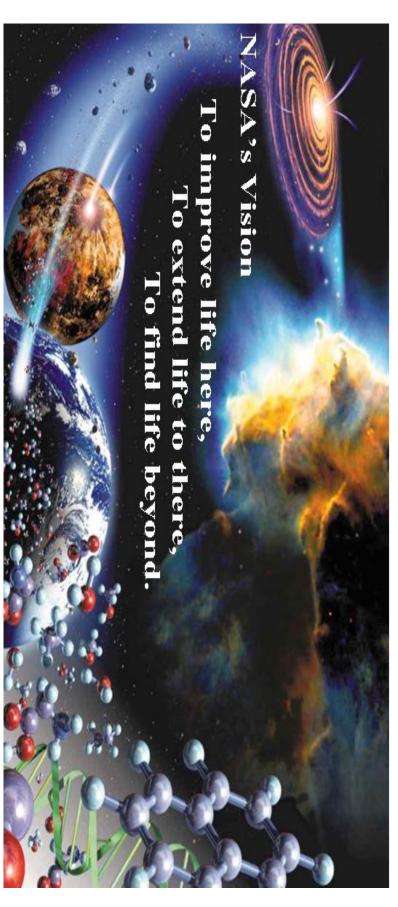
Dr. Gary T. Seng, Program Manager NASA Glenn Research Center November 18, 2002



Presentation Outline



- Mission, Programs & Goals
- Current Investment Areas
- Future Directions





The NASA Mission

To inspire the next generation of explorers To explore the Universe and search for life To understand and protect our home planet

... as only NASA can.

NASA Enterprises---From Strategic Plan to Programs

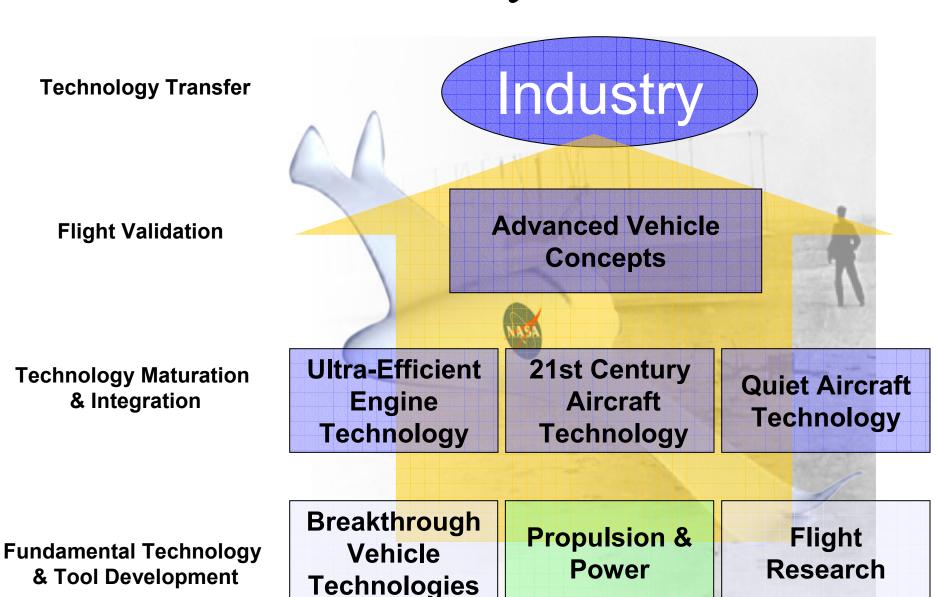




Commercial Technology Partnerships

Enterprise Objectives <u>In Progress - Draft</u>

Vehicle Systems



Propulsion & Power Program – Mission Statement



Provides leadership for the development of breakthrough aeronautics technologies to maintain U.S. technological and environmental superiority of propulsion and power systems

PROPULSION & POWER PROGRAM -Investment Areas & Projects

National Aeronautics and Space Administration
Glenn Research Center

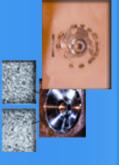
Turbine Engine Technologies



Smart Efficient Components



Oil-Free Turbine Engine Technology



UltraSafe Propulsion

New Propulsion Concepts



Pulse Detonation Engine Technology

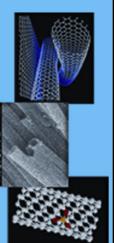


Revolutionary Aeropropulsion
Concepts



ZERO CO2 Emission Technologies

Foundation Technologies



ligh Temperature Nanotechnology

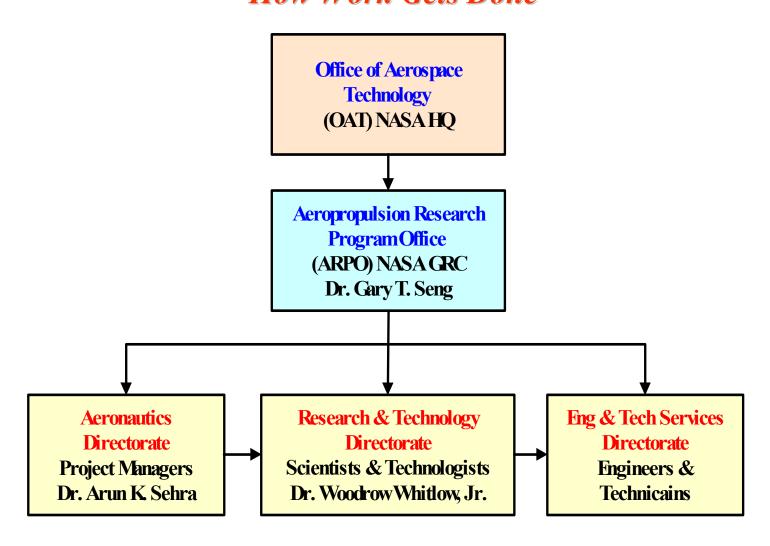


University Research,
Engineering & Technology
Institute (URETI)

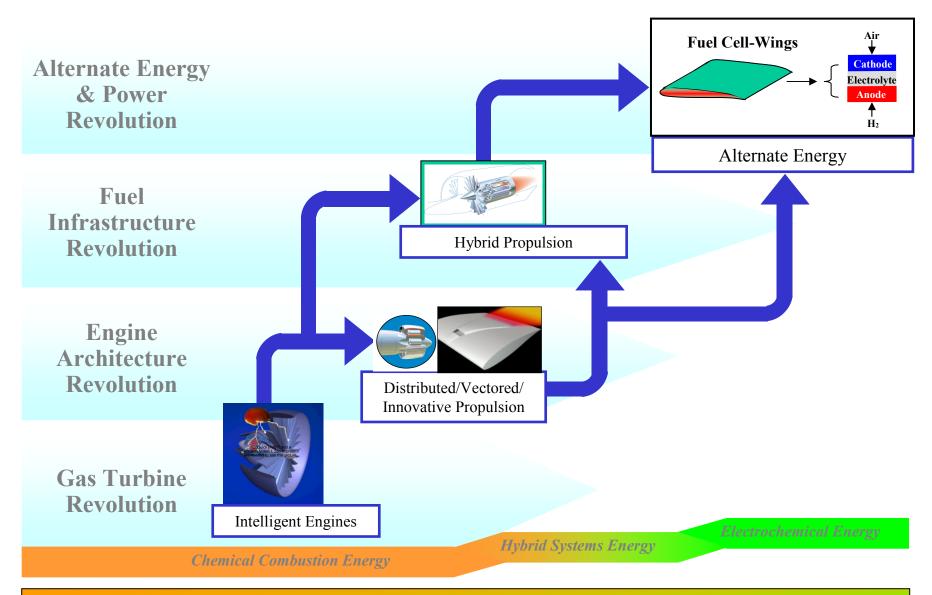


Higher Operating Temperature Propulsion Components

Program Flowdown to Matrix Organization "How Work Gets Done"



Aeropropulsion – NASA's Future Directions



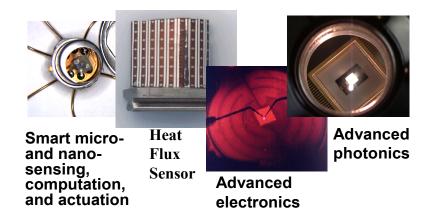
Foundation Technologies

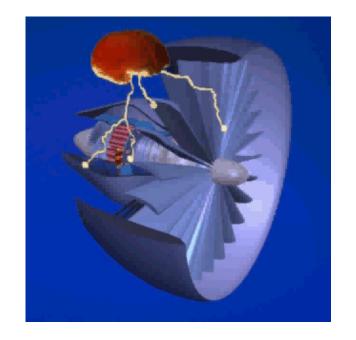
Intelligent Engine Technologies - Ultra-clean, Ultra-Quiet,

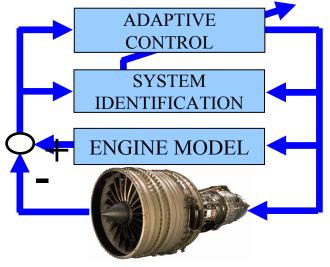
Intelligent Engine Attributes & Technologies

<u>Attributes</u> <u>Technologies</u>

- Sensing Embedded micro- and nanosensors
- Thinking Coupled simulation and data-feedback health and performance management
- Reacting Autonomic engine control strategies







Distributed, Vectored & Innovative Propulsion Systems

Distributed Engines/ Distributed Exhaust

 Multiple low-cost, low power engines deployed along wing providing distributed thrust and thrust vectoring

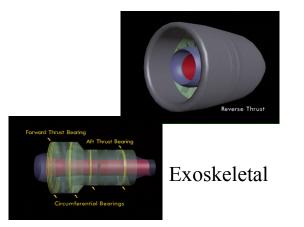
• Aircraft boundary layer ingestion using Micro-Turbines

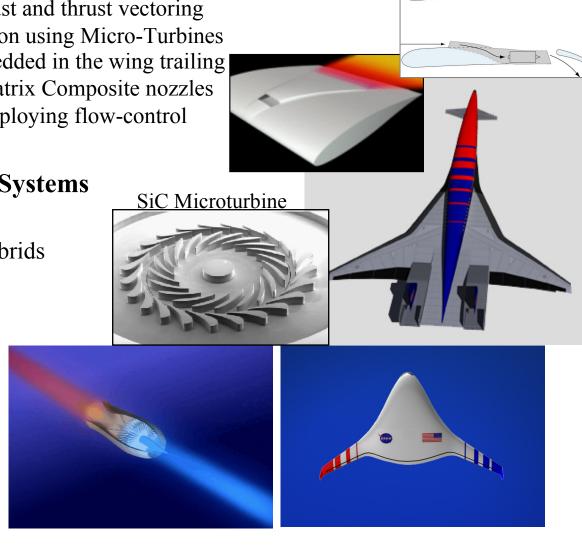
 High aspect-ratio nozzles embedded in the wing trailing edge using Ducted Polymer Matrix Composite nozzles

• Embedded inlets & nozzles employing flow-control

Revolutionary Propulsion Systems

- Exoskeletal Engine
- Pulse Detonation Engines/Hybrids
- Others

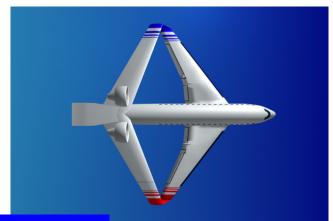


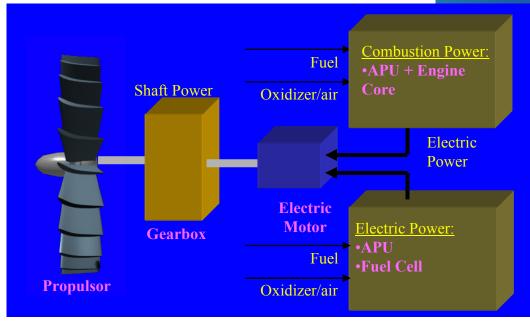


Alternative Energy Propulsion

Hybrid Combustion/Electric

- Takeoff thrust-augmenting Auxilliary Power Unit (APU)
- On-board electric power for zero emissions fan thrust





Propulsion & Power University Partnerships

Competitive Selections (FY02)

- ➤ 36 new **U-grants** = \$3.6M, open competition among universities
- > 9 new **U-contracts** = \$0.9M, open competition among universities and small businesses (\$2.0M total)
- ➤ Aeropropulsion & Power URETI Cooperative Agreement = \$3.0M/yr for up to 5 years

Other U-Investments

Project level grants, consortium support, faculty & student fellowships, and cooperative agreements = \$6.0M



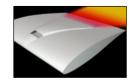


Summary



Propulsion& Power R&T will enable new engine and aircraft systems that will revolutionize aviation by providing:

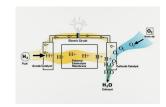
- Emerging ultra-low noise & emissions, using intelligent turbofans.
- Future distributed, vectored and/or innovative propulsion systems with 24 hour operations and greater community mobility.



• Successful hybrid combustion/electric propulsion systems leading to near-zero emissions and silent aircraft.

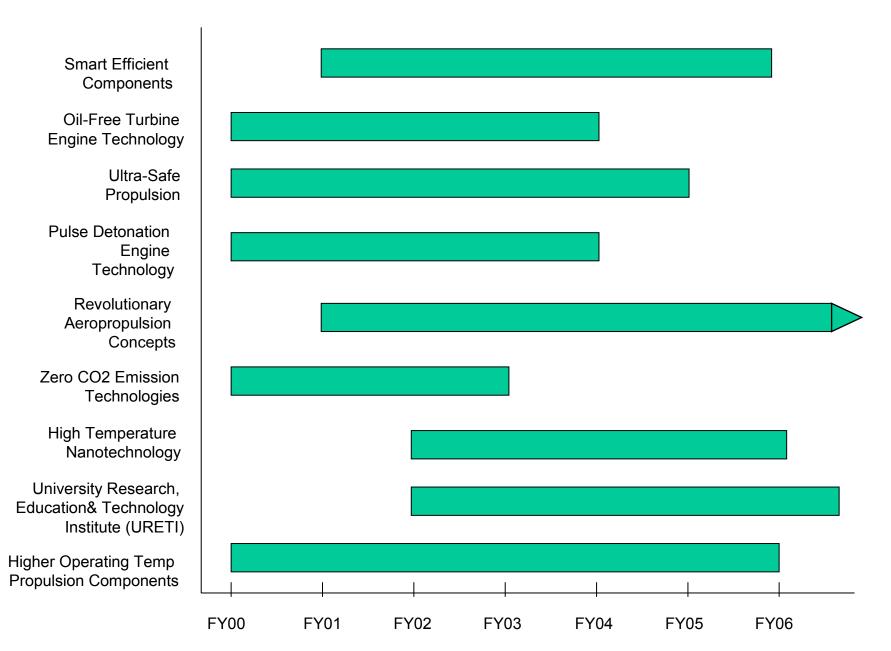


• The culmination of these revolutions will deliver all-electric powered airplanes, which achieve zero-impact emissions and noise with high-capacity, on-demand operation.



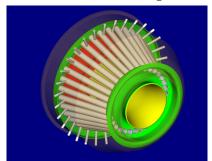
Back-up Slides

<u>Propulsion & Power – Project Timelines</u>



Current Efforts - New Propulsion Concepts

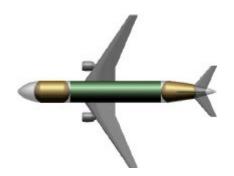
Pulse Detonation Engine Technology



Potential benefits:

- •20-30% combustor total pressure gain
- •22-26% improvement in fuel consumption Technical challenges:
- Materials fatigue
- Acoustics
- Unsteady performance
- Energy extraction
- More

Zero CO2 Emissions
Technologies



Revolutionary Aeropropulsion Concepts



Systems Analyses on Advanced Concepts

- Distributed/vectored propulsion
- Exo-skeletal
- Dual fan
- Levitated ducted fan Next Generation Fuel Cells Cryogenic Electric Motors

Next Generation Fuel Cells Alternatives for electric- powered flight Optimization of H2 fueled turbofan NOx emissions from H2 combustion

Current Efforts – Foundation Technologies

University Research, Engineering & Technology Institute (URETI)





Research Areas

System level engineering analysis and technology integration methods

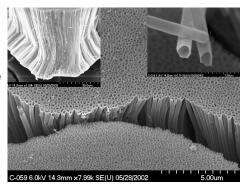
Enabling technologies for components and systems Intelligent engine components and systems High performance components Advanced power technologies

Higher Operating Temp Propulsion Components:

Extend temperature capability of all classes of materials throughout the entire engine.

- Develop life prediction capabilities for resulting materials and components.
- Validate material characterization behavior and component structural performance with data from rig/engine tests.

High Temperature Nanotechnology



Focus on Silicon Carbide (SiC) for high temp application :

- · high modulus
- hardness
- light weight
- oxidative stability
- · chemical resistance
- •H.T. semiconductor
- •H.T. optical material

Current Efforts - Turbine Engine Technologies

Silicon Carbide

Increase Compressor Stage Loading and Efficiency

Flow Control
Unsteady Aerodynamics/
Blade Flutter Control

Crack resistant

materials

Ultra-low Emission Combustors
Design/Analysis Tools

Lean Direct Injector (LDI)
National Combustor Code Development
Understanding of Combustion
Instabilities

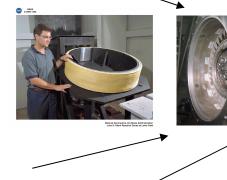
Increase Turbine Stage Loading and Efficiency

Develop Enabling

Sensors for Smart/Active Control Systems

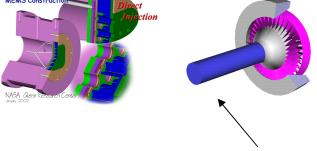
Turbine Heat Transfer Research/Modeling Flow Control for Low Pressure Turbines Low Pressure Turbine CFD Simulation

Light weight, high strength fan containment



Revolutionary Acoustic Seals
Advanced Non-Contacting Seals





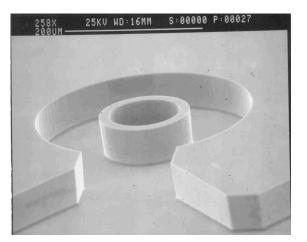
High-Temperature Magnetic Bearings Self-tuning Vibration Absorbers Air bearings to eliminate oil system



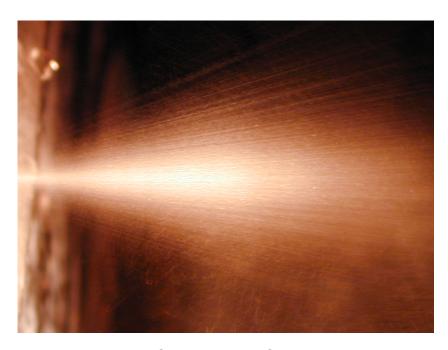
SEC Project - ADVANCED LEAN DIRECT INJECTOR CONCEPT



Quarter piece part of batch fabricated LDI laminate in silicon carbide



Flow channel fabricated in silicon



Evaluation of LDI nozzle fuel spray pattern for single fuel injector with mixing chamber

Accomplishments:

- •Completed proof of concept for assembly of the SiC laminate LDI array, including deep reactive ion etching batch fabrication and multi-layer wafer bonding
- •Completed initial thermal and structural analysis of the LDI assembly
- •Conducted single injector simplex test to evaluate LDI nozzle fluid spray pattern

SEC Project - ADVANCED LEAN DIRECT INJECTOR (LDI) CONCEPT

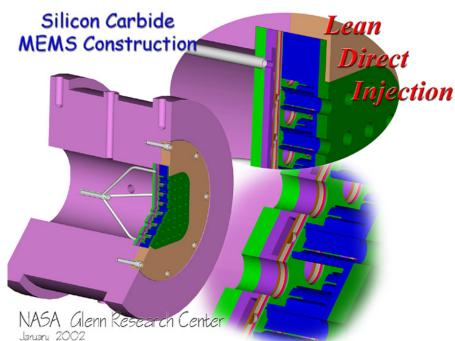
Objective:

Demonstrate revolutionary fuel injector concepts that utilize advanced technology, including metals, ceramics, and MEMS technology in flame-tube tests, to achieve the 80% NOx reduction goal, and reduce particulate and aerosol emissions.

January 2002



LDI Hardware installed in Flame Tube Test
Chamber

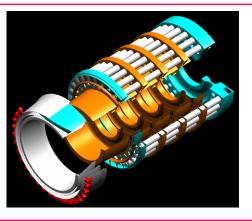


PDET Project - Hybrid Conceptual Design Study

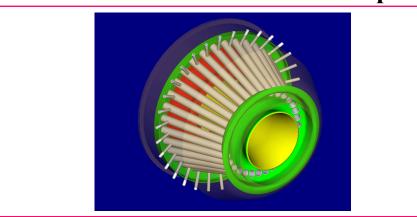
Objective: Complete assessment and conceptual design of PDE-based hybrid cycle and combined-cycle propulsion systems.

Shown: Representative combustor and engine concepts for PDE-based 50- and 300-passenger class subsonic vehicles and a Pulse Detonation Rocket Engine (PDRE) employed with a Rocket-Based Combined Cycle (RBCC) access-to-space vehicle.

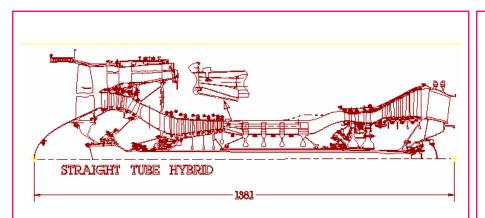
P&W PDE Combustor Concept



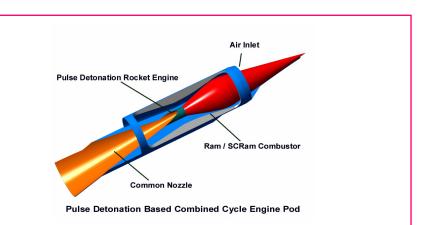
APRI PDE Combustor Concept



P&W Hybrid Engine Concept



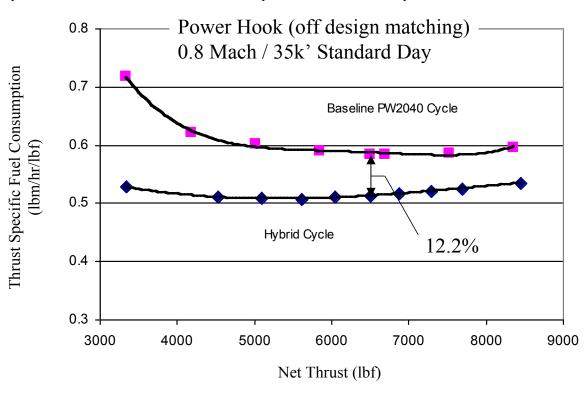
PDRE-RBCC Engine Concept



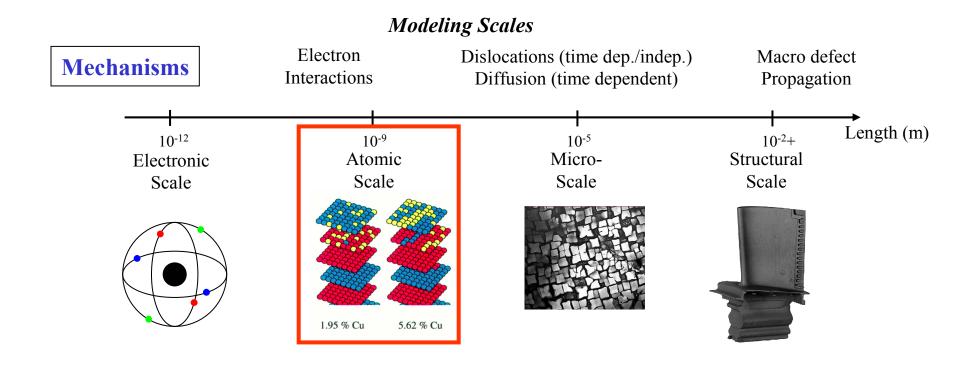
PDET Project - Pulse Detonation Engine (PDE)-Based System Conceptual Design

Results: A total of eleven design concepts were completed by the P&W, APRI and NASA study teams for hybrid (i.e., gas turbine engines) and combined-cycle propulsion systems utilizing pulse detonation (PD). First-order system performance analyses and hybrid engine component design and performance sensitivities were also completed. The hybrid engine concepts were designed to take best advantage of the potential benefits of a PD-based system. Initial assessment of the PDE concepts indicates increased performance potential.

- ➤ Fuel Economy (TSFC) at Cruise Improved 12.2%
- ▶6.3% lower engine weight
- ➤ 14.4% decrease in mission fuel burn



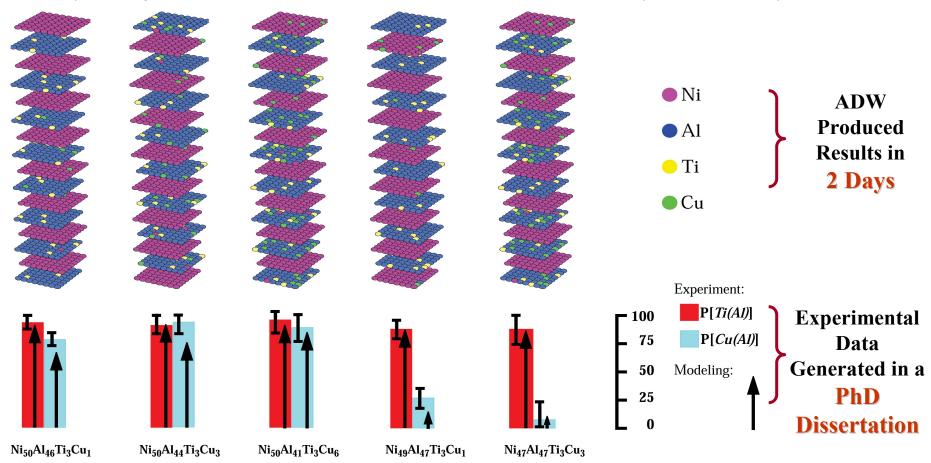
HOTPC Project - Computational Materials R&T



The Alloy Design Workbench Software Code will enable a designer/engineer to computationally create and evaluate a multitude of alloying materials, and assess the desired properties at a fraction of the cost and time that is associated with the current practices of building, testing, and analyzing.

HOTPC Project - Recent Model Results

Alloy Design Workbench Utilized to Model Site Occupancy in NiAl Alloys



Validated Results Generated in a Fraction of the Time